

PHYS2006

2011/2012  
Sem 1 Exam

(A1) 
$$\underline{R} = \frac{1}{M} \sum_{i=1}^N m_i \underline{r}_i \quad ; \quad M = \sum_{i=1}^N m_i \quad \left. \right\} [2]$$

where  $m_i$  are the masses and  $\underline{r}_i$  the position for the  $N$  particles. Thus

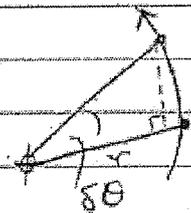
$$\underline{R} = \frac{1}{M} \sum_{i=1}^N m_i \dot{\underline{r}}_i = \frac{1}{M} \underline{P} \quad \text{or} \quad \boxed{\underline{P} = M \underline{R}} \quad \left. \right\} [2]$$

since the individual momenta are  $m_i \dot{\underline{r}}_i$

(A2) For a force  $\underline{F}$  applied at a point  $\underline{r}$  }  
 The torque  $\underline{\tau} = \underline{r} \times \underline{F}$  } (1) force subtended constant

$$\underline{\tau} = \frac{d}{dt} \underline{L} \quad \text{where } \underline{L} \text{ is angular momentum.} \quad \left. \right\} (1) [2]$$

If the force is central then  $\underline{F} \propto \underline{r}$ , so  $\underline{\tau} = 0 \Rightarrow \underline{L} = \text{constant w.r.t. time i.e. conserved} \quad \left. \right\} (1) [2]$

(A3)  In time  $\delta t$ , the particle's radius sweeps out an area  $\delta A = \frac{1}{2} r^2 \delta \theta$  } (1) given  
 therefore  $\frac{dA}{dt} = \frac{1}{2} r^2 \dot{\theta}$  } (1) [2]

but  $L = M r (r \dot{\theta}) = m r^2 \dot{\theta}$  is a constant, }  
 so  $\frac{dA}{dt}$  is a constant implying that equal

areas are swept out in equal times. } (2) areas swept out  
 This is Kepler's 2nd law. } (1) NB

(A4) Looking down from the N pole, the ball is projected tangentially when viewed from an inertial frame. Gravitational force is central so angular momentum is conserved. } (2) conserved  
 Therefore as ball falls, its angular velocity increases } (1)

(A4) continued -

with the consequence that the ball gets slightly ahead of the tower as it falls. [2]

(AS) The Coriolis force is the term

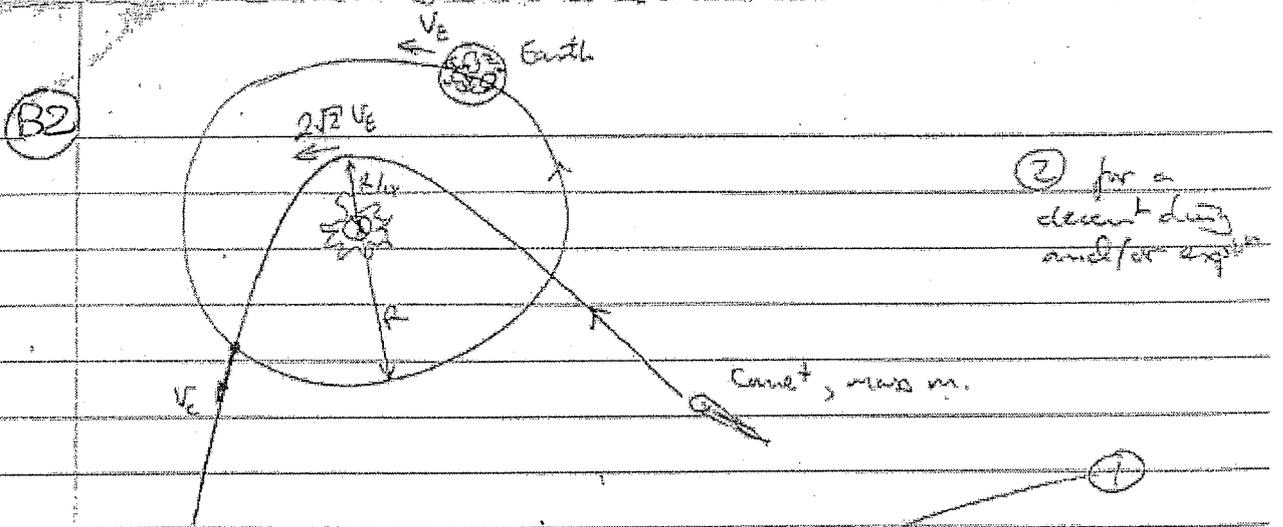
$$-2m \omega \times \underline{x}$$

that appears in Newton's equation of motion <sup>for the particle</sup> when considered in the rotating frame of the Earth. Here  $\omega$  is the angular velocity of the Earth, and  $m$  and  $\underline{x}$  are the mass and (apparent) velocity of the particle. [2]

It thus supplies a force at right angles to the direction of motion. Foucault's pendulum is simply a pendulum attached to a stationary frame, typically very large so that it can swing unaided for a long time. Then it is found <sup>(clockwise from above)</sup> that the plane in which it swings precesses at an angular velocity of  $\omega \sin \lambda$ ,  $\lambda$  being the latitude, as may be derived from the Coriolis force

For substantially correct [2] B





At closest approach comet's velocity is  $\perp$  to its position displacement from the sun therefore its angular momentum

$$L = m \left( \frac{R}{4} \right) (2\sqrt{2} v_E) = \frac{1}{\sqrt{2}} m R v_E \quad (*) \quad [5]$$

Its total energy  $E = \frac{1}{2} m (2\sqrt{2} v_E)^2 - \frac{GMm}{R/4} = 4m v_E^2 - 4 \frac{GMm}{R}$

When it crosses Earth's orbit with speed  $v_c$ ,  $E = \frac{1}{2} m v_c^2 - \frac{GMm}{R}$  (conservation energy) (2)

But Earth's motion  $\frac{v_E^2}{R} = \frac{GM}{R^2}$  (circular motion) (3)

$\therefore (1)(2) \Rightarrow 4v_E^2 - 3\frac{GM}{R} = \frac{1}{2} v_c^2$

$\therefore (3) \Rightarrow = v_E^2 \Rightarrow v_c = \sqrt{2} v_E \quad [7]$

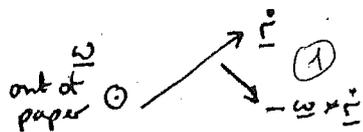
From (\*) we have that the tangential component of velocity when comet crosses Earth's orbit so  $v_t = v_E / \sqrt{2}$

$\therefore$  we have  $\cos \theta = \frac{1}{2} \Rightarrow \theta = 60^\circ$  i.e. angle at crossing  $\theta = 60^\circ$

B3

□ Two inertial force terms in eqn. motion.

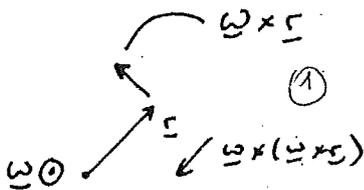
(i) Coriolis term  $-2m\omega \times \dot{\underline{r}}$  [2]



Direction of cross product shows that ②  
bug feels a force perp to its velocity,  
directed towards the right.

magnitude:  $2m\omega u$  [3]

(ii) centrifugal term.  $-m\omega \times (\omega \times \underline{r})$  [2]



$\omega \times (\omega \times \underline{r})$  points radially inwards, ②  
so  $-\omega \times (\omega \times \underline{r})$  is radially outwards

magnitude  $m\omega^2 r$  [3]

-normal "centrifugal force" term.

□ For bug not to slip: consider horizontal part of eqn. of motion. [2]

$\ddot{\underline{r}} = 0$  in the rotating frame so, [2]

$\underline{F} = 2m\omega \times \dot{\underline{r}} + m\omega \times (\omega \times \underline{r})$

o we clearly want  $|\underline{F}| < \mu mg$  for no-slip (using vertical part of eqn. motion) [2]

o Coriolis and centrifugal terms are orthogonal, so

$\mu^2 g^2 < 4\omega^2 u^2 + \omega^4 r^2$  [2]

So it can get to

$r = \frac{[\mu^2 g^2 - 4\omega^2 u^2]^{1/2}}{\omega^2}$

[2]  
before it slips

(B2) continued -

Comet can escape if it has <sup>zero or +ve</sup> energy at  $\infty$  distance from sun (= its kinetic energy there) but energy will be

$$E = 4mV_e^2 - \frac{4GMm}{R} \quad (\text{by } \textcircled{1})$$

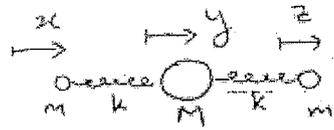
$$= 0 \quad (\text{by } \textcircled{2})$$

$\therefore$  Comet has just enough energy to escape  $\textcircled{1}$  [3]

(B1) Normal mode: each part of the system oscillates with same frequency, but in general different amplitude and phase.

[1]B

[1]B



[1]

$$MII \Rightarrow m \ddot{x} = -k(x-y) \quad [1]$$

$$M \ddot{y} = k(x-y) - k(y-z) \quad [1]$$

$$m \ddot{z} = k(y-z) \quad [1]$$

Thus

$$\begin{pmatrix} \ddot{x} \\ \ddot{y} \\ \ddot{z} \end{pmatrix} = \begin{pmatrix} -k/m & k/m & 0 \\ k/m & -2k/M & k/M \\ 0 & k/m & -k/m \end{pmatrix} \begin{pmatrix} x \\ y \\ z \end{pmatrix} \quad \begin{matrix} [2] \\ \text{(1 for matrix} \\ \text{1 for det)} \end{matrix}$$

Look for normal mode:  $\begin{pmatrix} x \\ y \\ z \end{pmatrix} = \begin{pmatrix} A \\ B \\ C \end{pmatrix} e^{i\omega t}$  [1]B

Require  $\begin{vmatrix} \omega^2 - k/m & k/m & 0 \\ k/m & \omega^2 - 2k/M & k/M \\ 0 & k/m & \omega^2 - k/m \end{vmatrix} = 0$  [2] (1 for det, 1 for ...)

$$\Rightarrow (\omega^2 - k/m) \{ (\omega^2 - 2k/M)(\omega^2 - k/m) - k^2/Mm \} - k^2/mM (\omega^2 - k/m) = 0 \quad [1]$$

$$\Rightarrow \omega^2 (\omega^2 - k/m) (\omega^2 - k/m - 2k/M) = 0 \quad [1]$$

\* Also for  $\omega = 0$ : find  $A=B=C \Rightarrow$  mode of uniform translation of whole system at const. velocity [1]\*

for intuitive derivation

$\omega = \sqrt{k/m}$ :  $\begin{pmatrix} 0 & k/m & 0 \\ k/m & k/m - 2k/M & k/M \\ 0 & k/m & 0 \end{pmatrix} \begin{pmatrix} A \\ B \\ C \end{pmatrix} = \begin{pmatrix} 0 \\ 0 \\ 0 \end{pmatrix}$   $B=0$   $A=-C$  [1]\*

Central mass at rest. Outer masses oscillate out of phase.

$\omega = \left( \frac{k}{m} + \frac{2k}{M} \right)^{1/2}$ :  $\begin{pmatrix} 2k/M & k/m & 0 \\ k/m & k/m & k/M \\ 0 & k/m & 2k/M \end{pmatrix} \begin{pmatrix} A \\ B \\ C \end{pmatrix} = \begin{pmatrix} 0 \\ 0 \\ 0 \end{pmatrix}$   $A=C$   $B = -\frac{2m}{M} C$  [1]

Outer two masses oscillate in phase; inner mass out of phase with outer masses, different amplitude. [1]

